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# International Journal of Solids and Structures

## Limitations in the design of deployable structures with straight scissors using identical elements

--Manuscript Draft--

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<b>Corresponding Author:</b>	Carlos Jose Garcia Mora, M.S. Sevilla University, Higher Technical School of Architecture of Sevilla (ETSAS) Sevilla, Sevilla SPAIN
<b>First Author:</b>	Carlos Jose Garcia Mora, M.S.
<b>Order of Authors:</b>	Carlos Jose Garcia Mora, M.S. Jose Sánchez-Sánchez, Professor of Architecture
<b>Abstract:</b>	After the recent publications on the design of deployable structures with straight scissors, it has been shown that the possibilities in the application of this type of structure are enormous: organic geometries, bistability control, etc. However, one of the aspects that have still not been studied is the particular case of using identical elements. On the one hand, the satisfaction of this geometric condition implies advantages in the structural behaviour and in the manufacturing and assembly process, but, on the other hand, this geometric constraint also limits design possibilities as will be demonstrated in this paper. Consequently, this research will develop a mathematical process to demonstrate that only non-bistable planes, non-bistable cylinders with a circle base and non-bistable spheres can be designed as deployable if straight scissors and identical elements are used.

# 1 Limitations in the design of deployable structures with 2 straight scissors using identical elements

Corresponding Author	
Full name	Carlos José García Mora
Name in scientific papers	Carlos J. García - Mora
Affiliation	Architecture School of Seville University, Spain (PhD candidate) Department of Building Structures and Geotechnical Engineering
Phone	+34 671 93 25 15
E-mail address	email@carlosjosegarciamora.com

3

Co-author	
Full name	Jose Sánchez Sánchez
Name in scientific papers	Jose Sánchez - Sánchez
Affiliation	Architecture School of Seville University, Spain (Professor) Department of Building Structures and Geotechnical Engineering

4

## 5 **Abstract**

6 After the recent publications on the design of deployable structures with straight  
7 scissors, it has been shown that the possibilities in the application of this type of  
8 structure are enormous: organic geometries, bistability control, etc. However, one of  
9 the aspects that have still not been studied is the particular case of using identical  
10 elements. On the one hand, the satisfaction of this geometric condition implies  
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16 deployable if straight scissors and identical elements are used.

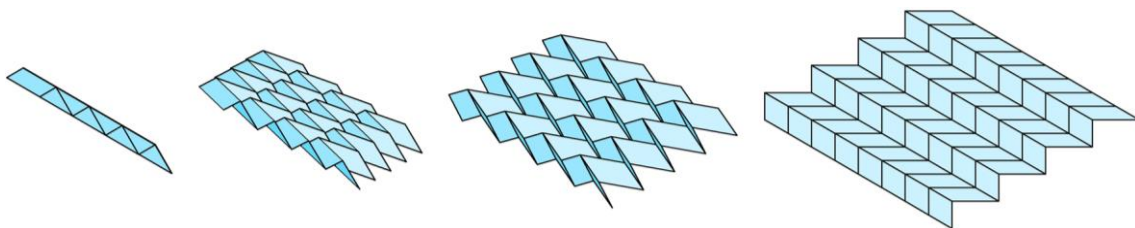
17 **Keywords:** Geometry; Deployable structure; Scissor; Mechanism; Folding; Kinematics;  
18 Identical element; Identical length; Identical cross section

19 **1. Introduction:**

20 Deployable structures are mechanisms that have the property to be folded in a  
21 compact module when its transport or storage is required and they will be deployed in  
22 the destination place. Due to the importance of this type of structures, numerous  
23 researchers and designers have developed novel methods to achieve a better  
24 prediction of the behaviour of the structure during the deployment process [1] [2] [3].

25

26 Although the world of deployable structures is enormous, a basic classification based  
27 on deployable structures of plates and deployable structures of elements can be  
28 proposed. The first group is composed of the deployable Origami (the deployment is  
29 obtained bending the surface) [4] [5] [6] and the deployable Kirigami (the deployment  
30 is obtained cutting the surface) [7] [8]. The most important property of these  
31 structures is that the joints are lines instead of points and the most famous example of  
32 deployable Origami is the Miura pattern where a flat shape can be folded in a small  
33 package (Fig. 1).

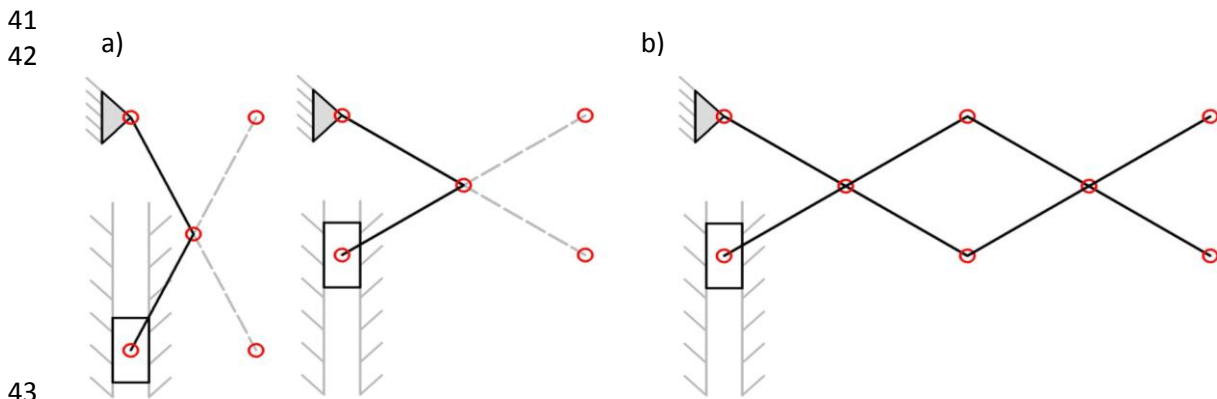


34  
35

**Fig. 1.** Deployment process of a flat Miura pattern.

36 The second group is composed of the deployable grid systems [9] [10] and the  
37 deployable scissor systems [11] [12] [13] (this paper will be focused on the scissor  
38 mechanisms). Basically, the scissors mechanisms are a crank mechanism [14] [15] with

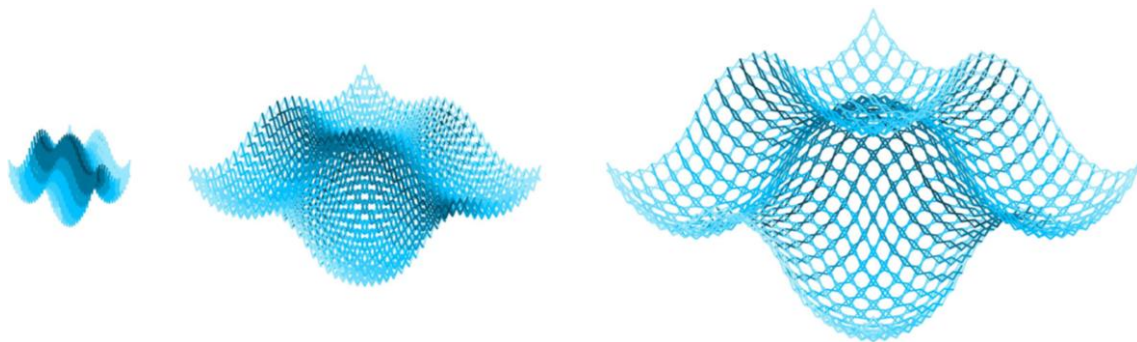
39 a geometric extension of the axes of all elements and this extension is going to be used  
40 to connect the next scissor and to get a transmission of the movement: Fig. 2



45 **Fig. 2.** (a) Basic crank mechanism; (b) Scissor mechanism based on a crank mechanism.

46 An application of this design strategy can be observed in Figure 3, where a deployable  
47 surface using translational units [16] [17] has been designed to cover a space. The  
48 deployment process of the structure is an important aspect that must be considered  
49 during the design process to evaluate possible collisions between the scissors.

50



51

52

53

**Fig. 3.** Deployment process of a surface using translational units.

54 Another parameter that is usually considered to classify the deployable structures  
55 with scissors is the bistability. In function of this property, two types of structures can  
56 be created: Bistable and non-bistable.

57

58 The main difference between both is the existence of a geometric incompatibility  
59 during the deployment process. In non – bistable structures, this incompatibility is null  
60 and there are no forces in the elements during the deployment process due to the  
61 elastic deformation. However, in bistable structures [18] [19] [20], there is a geometric  
62 incompatibility during the deployment process with an elastic deformation of the  
63 elements. This situation means that the structure is going to have two positions of  
64 stability: the folded position and the unfolded position.

65

66 Despite of the wide range of design possibilities (bistability, angle between the  
67 elements [21] [22] [23] [24], etc.), a question that has not still answered in the  
68 deployable structure world is the following: Which types of deployable structures can  
69 be obtained if the length of all elements is identical? and, are there infinite design  
70 possibilites or just a few design options?. The interest in this geometrical constraint is  
71 because this property can provide the following advantages:

72

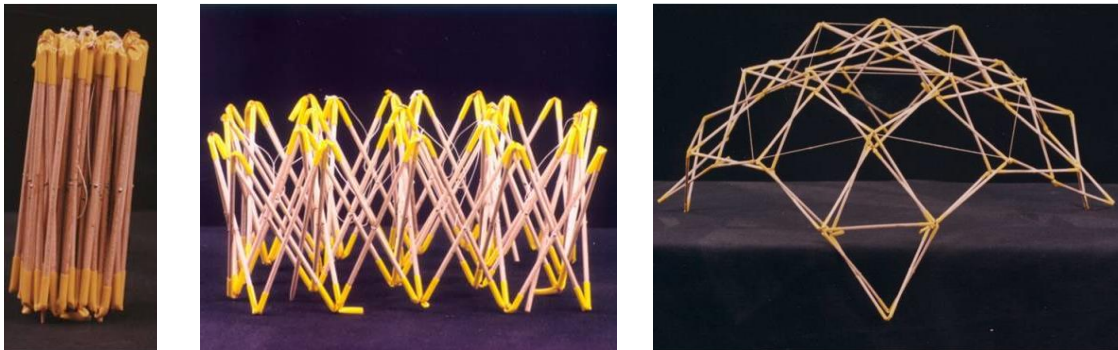
73 a) The natural frequencies of the structure will be higher: the use of elements with an  
74 identical length will allow the creation of deployable structures with a higher stiffness  
75 and, in consequence, with less horizontal displacements.

76 b) The manufacturing and construction process may be simpler when the length of the  
77 elements is a critical point: the worker does not need to change the cut length of the  
78 elements and its transport would be easier. However, the position of the middle joint  
79 will be different in each element and, in consequence, necessary measures must be  
80 taken in order to avoid an incorrect orientation of the elements.

81

82 Two designers who have already proposed this type of structures were Félix Escrig  
83 Pallarés and Jose Sánchez Sánchez. These authors manufactured a certain quantity of  
84 elements with the same length and with an excentricity in the middle joint. The  
85 connection of these elements gave the final geometry. A result of their works can be  
86 observed in Fig. 4, where a sphere with elements of the same length has been  
87 designed (the cables are attached after the deployment process).

88



89

90 **Fig. 4.** Model of a sphere with elements of the same length and with cables to get  
91 rigidization.

92 Consequently, the goal of this paper will be to obtain the bistable and non-bitable  
93 geometries that can be designed as deployable if the next condition is established: The  
94 length of all elements must be the same in all scissors of the structure. It is important  
95 to highlight that the work of this research is not a method to design deployable  
96 structures and what is going to be developed here are the shapes that can be created  
97 having in consideration this geometric constraint.

98

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100

101

102 **2. Methodology:**

103

104 The methodology that is going to be developed in this research is the following:

105

106 - Step 1: The geometric constraints that scissors with same length elements must  
107 satisfy will be imposed. This section is going to be called: "Mathematical  
108 development".

109

110 - Step 2: The geometric constraints of step 1 will be applied to a curve and the limits in  
111 the design of the deployable curve will be obtained. This section is going to be called:  
112 "Application to a curve".

113

114 - Step 3: The geometric constraints of step 1 will be applied to a surface and the limits  
115 in the design of the deployable surface will be obtained. This section is going to be  
116 called: "Application to a surface".

117

118 - Step 4: A comparison between a deployable structure with identical elements and a  
119 deployable structure with different length of the elements will be developed. This  
120 section is going to be called: "Results". The goal is to discover which advantages and  
121 disadvantages have the use of identical elements in terms of structural behaviour  
122 (vertical deformations and natural frequencies).

123

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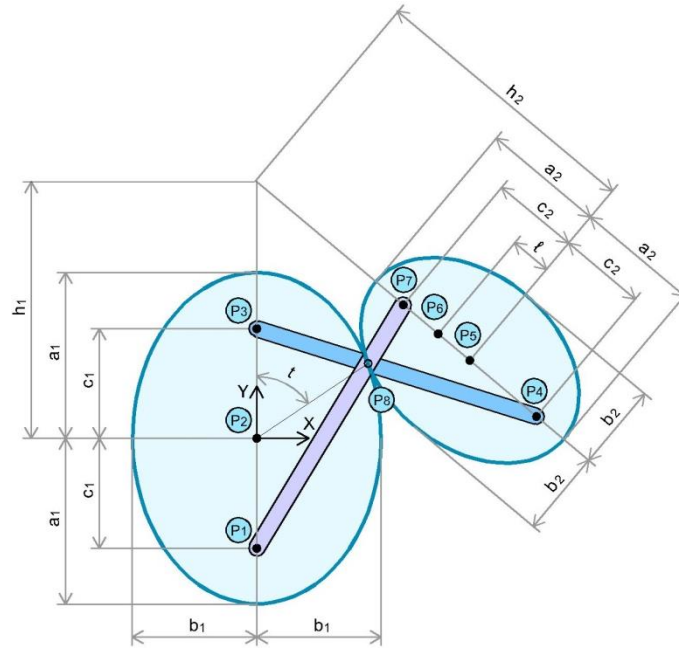
126 **3. Mathematical development:**

127

128 The situation of geometric convergence where the cross point between the elements

129 is in the tangency point between the ellipses has been represented in Fig. 5.

130



131

132

133

**Fig. 5.** Convergence situation for two ellipses in the plane.

134

$$P_1 = (0, -c_1) \quad (1)$$

135

$$P_2 = (0, 0) \quad (2)$$

136

$$P_3 = (0, c_1) \quad (3)$$

137

138 Where:

139

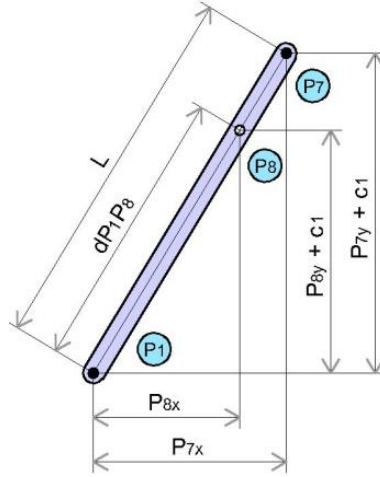
$$k = \sqrt{\left[\frac{\cos(t)}{a_1}\right]^2 + \left[\frac{\sin(t)}{b_1}\right]^2} \quad \text{with } 0 \leq t \leq 2 \cdot \pi \quad (5)$$

140

141 In addition, the geometric conditions of Fig. 6 and Fig. 7 must be satisfied:

142

143



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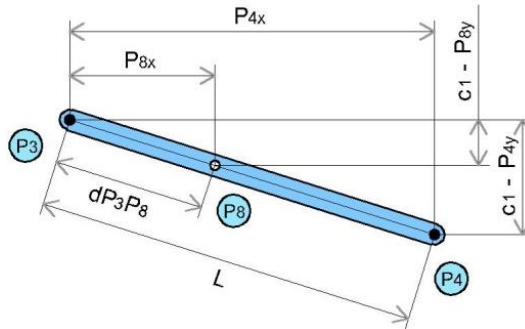
**Fig. 6.** Distance between points in the element  $P_1P_8P_7$ .

$$\frac{P_{7x}}{P_{8x}} = \frac{L}{dP_1P_8} \rightarrow P_{7x} = P_{8x} \cdot \frac{L}{dP_1P_8} \quad \text{with } L = \text{Length of the element } P_1P_8P_7 \quad (6)$$

146

$$\frac{P_{7y} + c_1}{P_{8y} + c_1} = \frac{L}{dP_1P_8} \rightarrow P_{7y} = (P_{8y} + c_1) \cdot \frac{L}{dP_1P_8} - c_1 \quad \text{with } L = \text{Length of the element } P_3P_8P_4 \quad (7)$$

147



148

149

**Fig. 7.** Distance between points in the element  $P_3P_8P_4$ .

$$\frac{P_{4x}}{P_{8x}} = \frac{L}{dP_3P_8} \rightarrow P_{4x} = P_{8x} \cdot \frac{L}{dP_3P_8} \quad (8)$$

150

$$\frac{c_1 - P_{4y}}{c_1 - P_{8y}} = \frac{L}{dP_3P_8} \rightarrow P_{4y} = (P_{8y} - c_1) \cdot \frac{L}{dP_3P_8} + c_1 \quad (9)$$

151

152 Consequently:

153

$$P_4 = \left( P_{8x} \cdot \frac{L}{dP_3P_8}, (P_{8y} - c_1) \cdot \frac{L}{dP_3P_8} + c_1 \right) \quad (10)$$

154

$$P_7 = \left( P_{8x} \cdot \frac{L}{dP_1P_8}, (P_{8y} + c_1) \cdot \frac{L}{dP_1P_8} - c_1 \right) \quad (11)$$

155

156 Then:

157

$$P_5 = \left( \frac{P_{4x} + P_{7x}}{2}, \frac{P_{4y} + P_{7y}}{2} \right) \quad (12)$$

158

$$P_6 = (P_{5x} + \Delta l_x, P_{5y} + \Delta l_y) = \left( \frac{P_{4x} + P_{7x}}{2} + \Delta l_x, \frac{P_{4y} + P_{7y}}{2} + \Delta l_y \right) \quad (13)$$

159

160 On the other hand:

161

$$dP_3 P_8 = \sqrt{\left[ \frac{\sin(t)}{k} \right]^2 + \left[ \frac{\cos(t)}{k} - c_1 \right]^2} \quad (14)$$

162

$$dP_1 P_8 = \sqrt{\left[ \frac{\sin(t)}{k} \right]^2 + \left[ \frac{\cos(t)}{k} + c_1 \right]^2} \quad (15)$$

163

164 Finally, if Eq. (10), Eq. (11), Eq. (12), Eq. (13), Eq. (14) and Eq. (15) are combined, the

165 equation of the convergence curve is obtained: Eq. (16) and Eq. (17).

166

$$x(t) = P_{6x}(t) = \frac{n_1 \cdot L}{2} \cdot \left[ \frac{1}{\sqrt{n_1^2 + (n_2 - c_1)^2}} + \frac{1}{\sqrt{n_1^2 + (n_2 + c_1)^2}} \right] + \Delta l_x \quad (16)$$

167

$$y(t) = P_{6y}(t) = \frac{L}{2} \cdot \left[ \frac{n_2 - c_1}{\sqrt{n_1^2 + (n_2 - c_1)^2}} + \frac{n_2 + c_1}{\sqrt{n_1^2 + (n_2 + c_1)^2}} \right] + \Delta l_y \quad (17)$$

168

169 Where:

170

$$n_1 = \frac{\sin(t)}{k} \quad (18)$$

171

$$n_2 = \frac{\cos(t)}{k} \quad (19)$$

172

173 The most common design case that can be found is having the centres of all ellipsoids

174 in the surface that is going to be designed as deployable ( $P_5$  will belong to the surface).

175 To achieve that, the following values shall be considered:  $l = 0 \rightarrow \Delta l_x = 0$  and  $\Delta l_y = 0$ .

176 The parameters that can be controlled by the designer are:  $a_1$ ,  $b_1$  and  $L$  ( $c_1$  is a function

177 that depends on  $a_1$  and  $b_1$ ). Likewise, the  $L$  parameter is obtained by multiplying Eq.

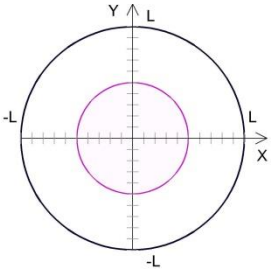
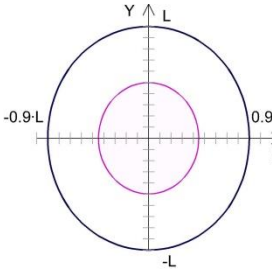
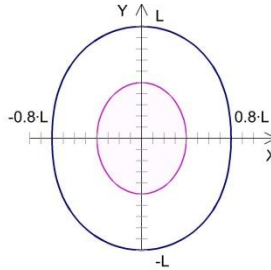
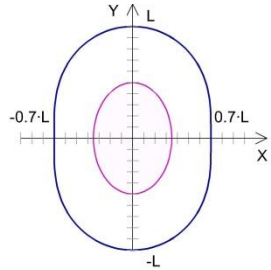
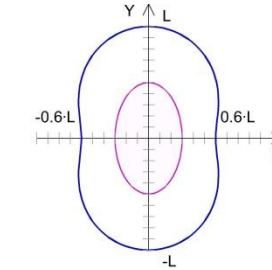
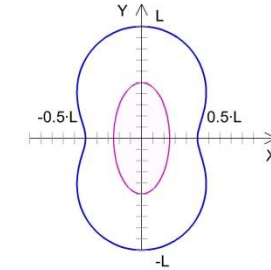
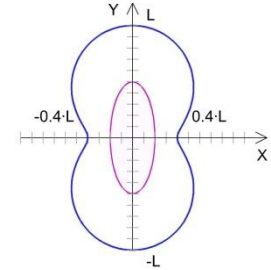
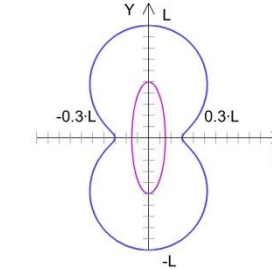
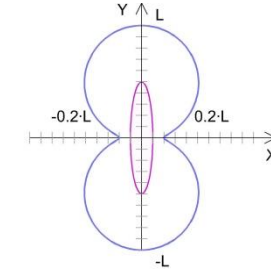
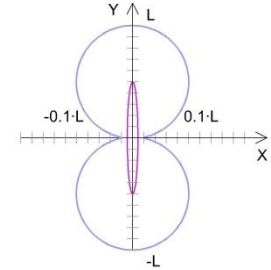
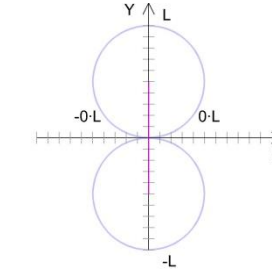
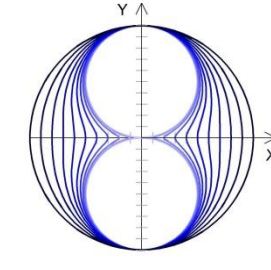
178 (16) and Eq. (17). Consequently, the  $L$  parameter is going to influence the size of the

179 convergence curve but not its shape. This situation means that the parameters that

180 have an influence on the shape of the curve are just  $a_1$  and  $b_1$ . The relationship

181 between  $a_1$ ,  $b_1$  and the shape of the convergence curve for  $l=0$  can be seen in Table 1.

182

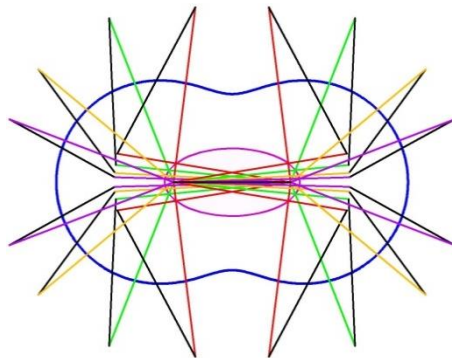
$b_1/a_1$	1	0.9	0.8
Convergence curve			
$b_1/a_1$	0.7	0.6	0.5
Convergence curve			
$b_1/a_1$	0.4	0.3	0.2
Convergence curve			
$b_1/a_1$	0.1	0	[1,0]
Convergence curve			

183

184

**Table 1.** Evolution of the convergence curve in function of  $b_1/a_1$  for  $l = 0$ .

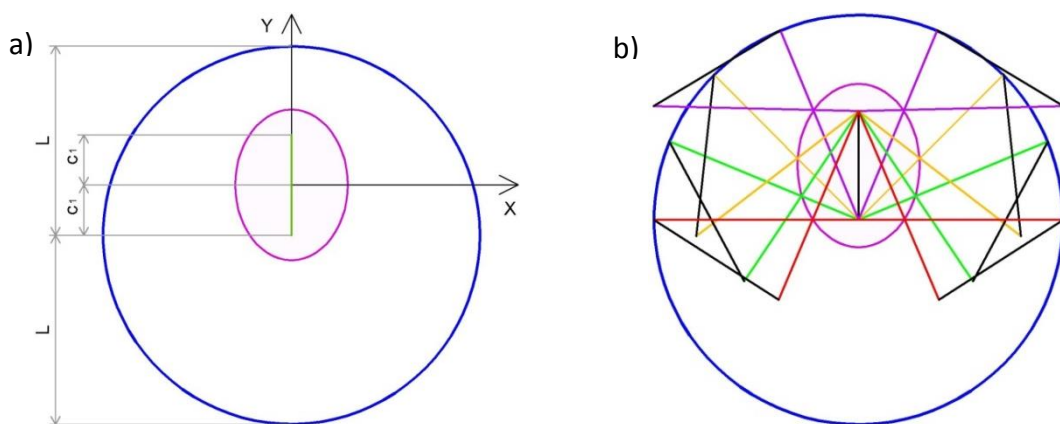
185 In the following image, an example of some scissors that belong to the convergence  
 186 curve can be observed (Fig. 8):



187  
 188 **Fig. 8.** 12 scissors obtained in the convergence curve for  $l = 0$  and for  $b_1/a_1 = 0.5$   
 189 (elements of same colour belong to the same scissor) (elements of all scissors have the  
 190 same length).

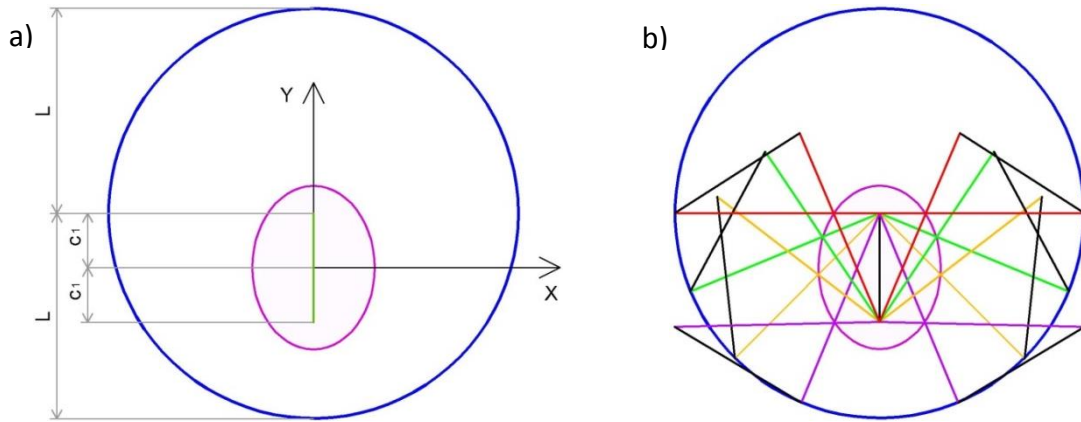
191 However, the previous design condition has a considerable disadvantage: the final  
 192 shape will not be the original surface if a textile is hold using the superior or inferior  
 193 joints of the structure. To solve that, it is necessary that the superior joints ( $P_3$  and  $P_7$ )  
 194 or the inferior joints ( $P_1$  and  $P_4$ ) belong to the original surface. In terms of equations:

195 - For the first case (superior joints), the next conditions are mandatory:  $l = c_2 \rightarrow \Delta l_x =$   
 196  $P_{7x} - P_{5x}$  and  $\Delta l_y = P_{7y} - P_{5y}$  (Fig. 9)



197  
 198 **Fig. 9.** (a) Convergence curve for  $l = c_2$  (blue circle with centre in the bottom focus of  
 199 the original ellipse); (b) 8 scissors that belong to the convergence curve for  $l = c_2$

200 - For the second case (inferior joints), the next conditions are mandatory:  $l = -c_2 \rightarrow \Delta l_x =$   
 201  $P_{4x} - P_{5x}$  and  $\Delta l_y = P_{4y} - P_{5y}$  (Fig. 10)



202  
 203 **Fig. 10.** (a) Convergence curve for  $l = -c_2$  (blue circle with centre in the top focus of the  
 204 original ellipse); (b) 8 scissors that belong to the convergence curve for  $l = -c_2$

205 When the convergence curve is obtained for any of the previous cases, the scissor in  
 206 one point of this curve is going to be defined by 2 parameters: orientation ( $h_1$ ) and the  
 207 value of the focal distance in the point of the convergence curve ( $c_2$ ):

208

209 a) Orientation =  $h_1$  (for any value of  $l$  parameter):

210

211 The line that is defined by  $P_4$  and  $P_7$  is:

212

$$y = \frac{P_{4y} - P_{7y}}{P_{4x} - P_{7x}} \cdot x + P_{7y} - P_{7x} \cdot \frac{P_{4y} - P_{7y}}{P_{4x} - P_{7x}} \quad (20)$$

213

214 In addition,  $h_1 = y(x = 0)$ . If this condition is replaced in Eq. (20), Eq. (10) and Eq. (11),

215 the  $h_1$  equation is obtained.

216

217

218

219

220

221 b) Focal distance =  $c_2$  (for any value of  $l$  parameter):

222 The next equation is obtained from Fig. 5:

$$c_2 = dP_7P_5 = \sqrt{(P_{7x} - P_{5x})^2 + (P_{7y} - P_{5y})^2} = \sqrt{\left(\frac{P_{7x} - P_{4x}}{2}\right)^2 + \left(\frac{P_{7y} - P_{4y}}{2}\right)^2} \quad (21)$$

223

224 If Eq. (10) and Eq. (11) are replaced in Eq. (21), the equation of focal distance is

225 obtained. Equations of orientation ( $h_1$ ) and focal distance ( $c_2$ ) will be used in the

226 Chapter 5: Application to a surface.

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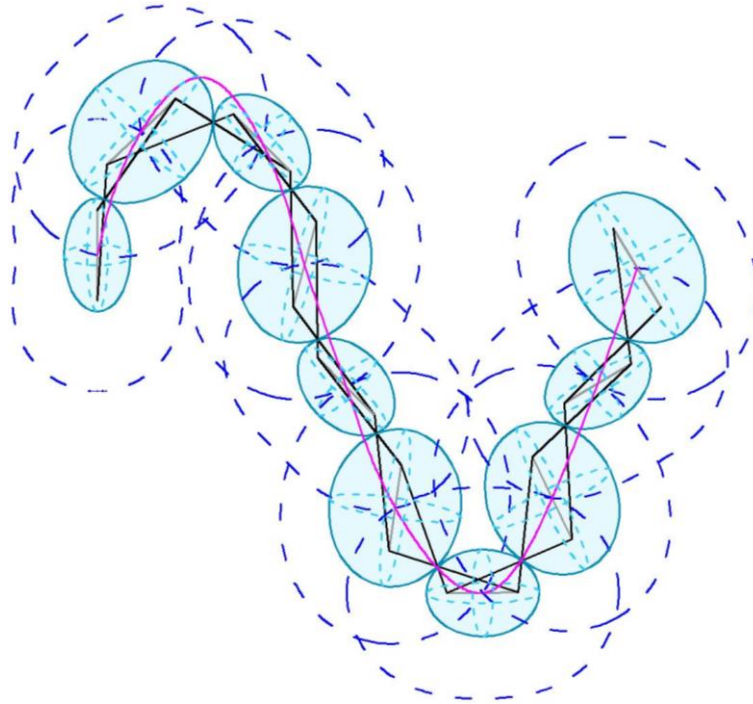
262

263 **4. Application to a curve:**

264

265 The first application case that is going to be solved is the situation with  $l = 0$  (the centre

266 of all ellipsoids is in the curve that is going to be designed as deployable) (Fig. 11).



267

268

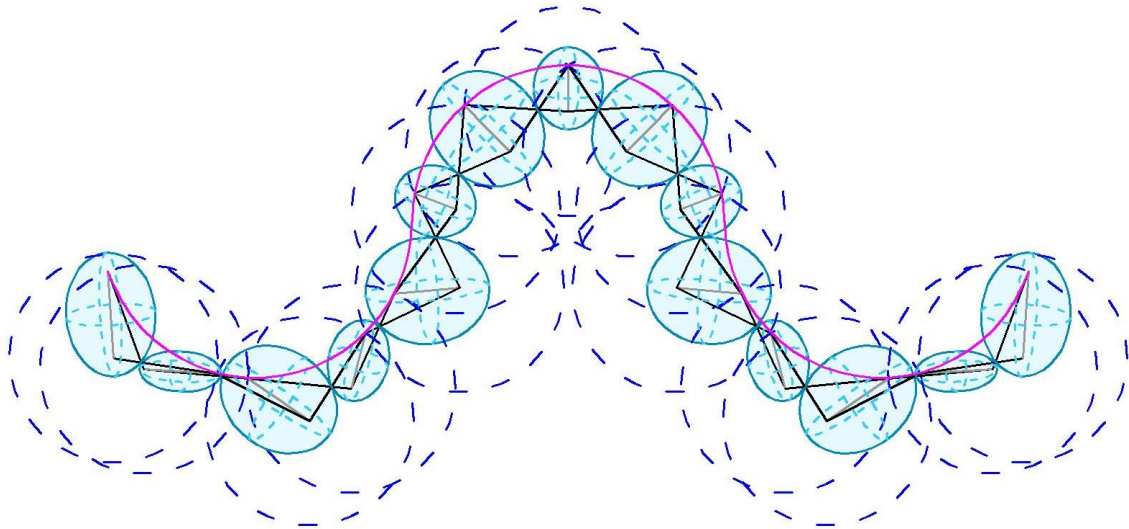
269 **Fig. 11.** Deployable structure using an identical length in all elements and with  $l = 0$

270 (purple curve = original curve; discontinuous blue curve = convergence curve).

271 The second case that will be developed is the situation with  $l = c_2$  (the top point of all

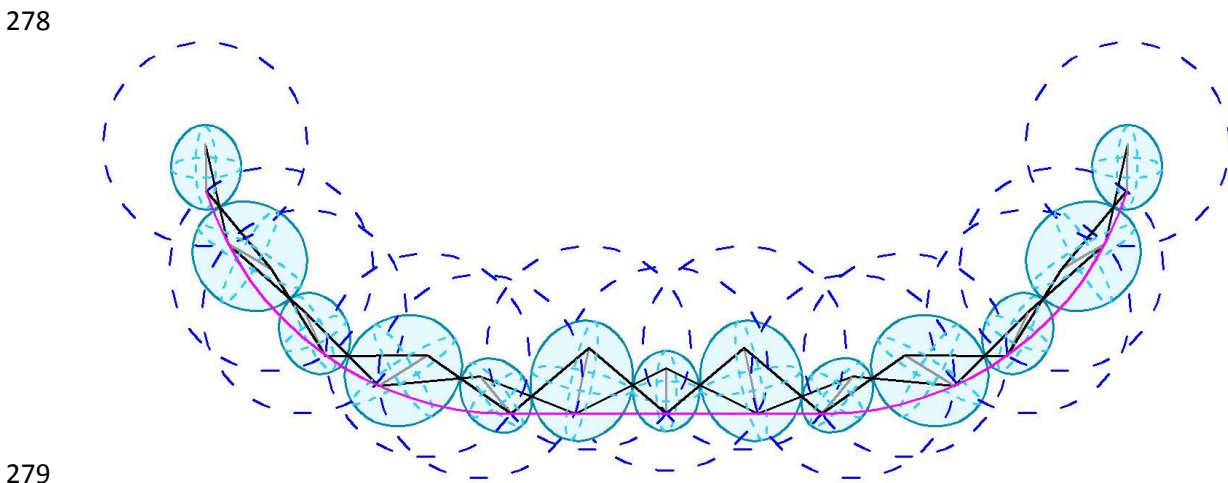
272 scissors is going to be in the curve that is going to be designed as deployable) (Fig. 12).





273  
 274 **Fig. 12.** Deployable structure using an identical length in all elements and with  $l = c_2$   
 275 (purple curve = original curve; discontinuous blue curve = convergence curve).

276 The last case is the situation with  $l = -c_2$  (the bottom point of all scissors is going to be  
 277 in the curve that is going to be designed as deployable) (Fig. 13).



279  
 280  
 281 **Fig. 13.** Deployable structure using an identical length in all elements and with  $l = -c_2$   
 282 (purple curve = original curve; discontinuous blue curve = convergence curve).

283  
 284  
 285

286 **5. Application to a surface:**

287

288 The application of this mathematical development to a surface is more complex and, in  
289 consequence, the following considerations are going to be taken:

290

291 a) When a curve is designed as deployable, the designer always works using one  
292 ellipsoid (or ellipse in case of a flat curve). However, when a surface is designed as  
293 deployable, the designer always works using two ellipsoids simultaneously in the  
294 space. The equation of  $c_2$  gives a different value for each point of the convergence  
295 surface and when a surface is designed as deployable,  $c_2$  must simultaneously have the  
296 same value between both original ellipsoids. This situation can only happen in two  
297 cases:

298

299 a1) Both original ellipsoids are symmetric or they are obtained using a rotation.  
300 In this case, the convergence surfaces will be symmetric and the intersection  
301 between them will give a curve where all of its points will have the same  $c_2$   
302 value between both ellipsoids. The problem is that these geometric conditions  
303 hugely limit the design possibilities.

304

305 a2) The existence of a relationship between input parameters that allows the  
306 creation of a convergence surface where all points will have the same  $c_2$  value.  
307 This assumption implies that:

308

$$c_2(t = t_i) = c_2(t = t_{i+1}) \quad (22)$$

309

310

Eq. (22) must be satisfied in all points of the convergence surface. In

311

consequence, Eq. (23) can be defined:

312

$$c_2(t = 0^\circ) = c_1 = c_2(t = 90^\circ) = L \cdot \frac{c_1}{a_1} - c_1 \quad (23)$$

313

314

Then:

315

$$c_1 = L \cdot \frac{c_1}{a_1} - c_1 \rightarrow L = 2 \cdot a_1 \quad (24)$$

316

317

If Eq. (24) is satisfied:

318

$$c_2(L = 2 \cdot a_1 \text{ for any value of } t) = c_1 \quad (25)$$

319

320

Consequently, if the length of all elements is equal to  $2 \cdot a_1$ , the  $c_2$  value is going

321

to be constant in all points of the convergence surface for any orientation of

322

both original ellipsoids.

323

324

Also, an aspect that is important to highlight is that the expression of  $c_2$  does

325

not depend on the "l" parameter and, consequently, Eq. (24) and Eq. (25) are

326

going to be satisfied for any value of "l". Finally, the following relationships are

327

satisfied:

328

$$dP_3P_8 + dP_1P_8 = 2 \cdot a_1 = L = dP_3P_8 + dP_4P_8 = dP_1P_8 + dP_7P_8 \quad (26)$$

329

$$dP_7P_8 + dP_4P_8 = 2 \cdot a_2 \quad (27)$$

330

331

If Eq. (27) is replaced in Eq. (26):

332

$$2 \cdot L - 2 \cdot a_2 = L \rightarrow L = 2 \cdot a_2 \rightarrow a_1 = a_2 \rightarrow b_1 = b_2 \rightarrow c_1 = c_2 \quad (28)$$

333

334

335

336 b) Not only  $c_2$  value has to be the same between both original ellipsoids, but also the  
 337 orientation of  $c_2$  between both ellipsoids must be the same. This situation means that  
 338  $h_1$  value between both original ellipsoids must be the same. To study this situation,  
 339 two cases can be found:

340

341 b1)  $h_1$  value will be the same between both original ellipsoids if the ellipsoids  
 342 are symmetric or if they have a relationship of a rotation. However, these  
 343 geometric conditions hugely limit the design possibilities.

344

345 b2) The existence of a relationship between input parameters that allows the  
 346 creation of a convergence surface where all points will have the same  $h_1$  value.  
 347 This assumption implies that:

348

$$h_1(t = t_i) = h_1(t = t_{i+1}) \quad (29)$$

349

350 Eq. (29) must be satisfied in all points of the convergence surface. In  
 351 consequence, Eq. (30) can be defined:

352

$$h_1(t = 0^\circ) = L - a_1 = h_1(t = 90^\circ) = c_1 \cdot \left[ \frac{L}{a_1} - 1 - \frac{2 \cdot \left(1 - \frac{L}{a_1}\right)}{1 - 1} \right] = \infty \quad (30)$$

353

354 The next step is the study of the lateral limits in the previous equation:

355

$$h_1(t = 90^+) = -\infty \quad \text{and} \quad h_1(t = 90^-) = +\infty \quad (31)$$

356

357 The lateral limits are different and, in consequence, the equation does not  
 358 converge either in  $+\infty$  or  $-\infty$ . Consequently, there is not a relationship  
 359 between the input parameters that allows the existence of a convergence  
 360 surface where all of their points have the same  $h_1$  value.

361

362 c) The study will be done for  $l = c_2$ . "l" is a parameter that allows the control of the  
363 tessellation, but it does not guarantee the geometric convergence. Consequently, the  
364 following study will give the same results for any value of "l" ( $l = \text{constant}$ ,  $l = c_2$  or  $l = -$   
365  $c_2$ ). On the other hand, the use of  $l = c_2$  has the following mathematical advantages:

366

367 c1) The intersection between both convergence surfaces will be the  
368 intersection between two spheres. This situation means that the intersection  
369 curve will be a circle.

370

371 c2) A sphere has a position in the space but not an orientation. Consequently,  
372 the use of a variable to modify the orientation of the convergence surface is not  
373 necessary.

374

375 d) The angular orientation between both original ellipsoids has an influence on the size  
376 of the convergence curve but not on the solution of the convergence (the angular  
377 position only implies a rotation and not a displacement).

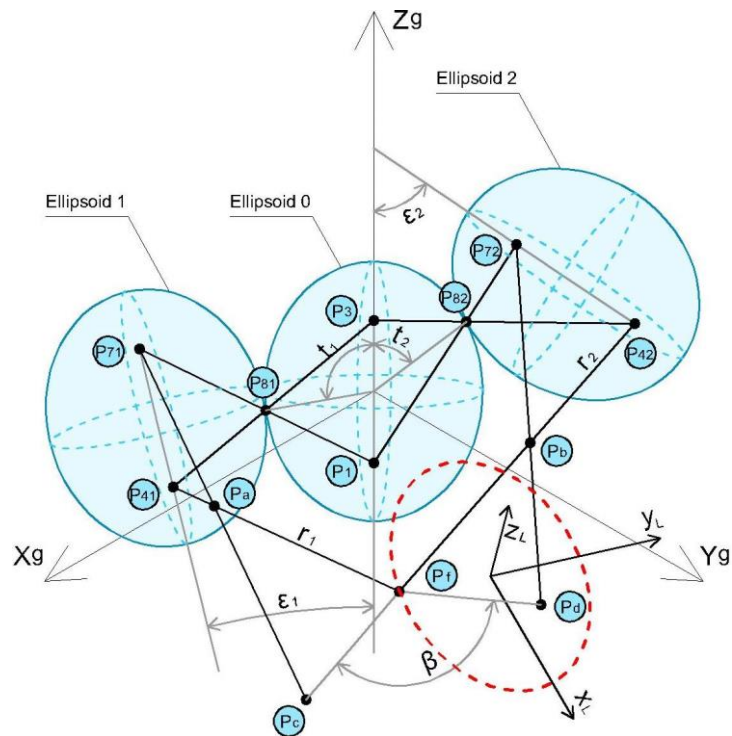
378

379 Once the previous conditions have been established, the goal is to find the set of  
380 points in the space that give a scissor with the same orientation between both original  
381 ellipsoids (the nonexistence of a mathematical relationship between both original  
382 ellipsoids that allows a convergence surface with the same  $h_1$  value has been  
383 demonstrated before).

384

385 This analysis is going to be done using 3 ellipsoids with a rotation of  $90^\circ$  (the results of  
 386 this study with another angle will be the same due to the relationship of rotation and  
 387 not of translation). As has been established before, the study will use  $l = c_2$  and the  
 388 centres of the spheres will be in  $P_{41}$  and  $P_{42}$ . The intersection between the convergence  
 389 surfaces will give the convergence curve (the red and discontinuous curve in the next  
 390 figure). The final step is to obtain the scissor from each original ellipsoid and the angle  
 391 between the focal distances of the scissors. When this angle is 0, the orientation value  
 392 between both original ellipsoids will be the same (Fig. 14).

393



394

395

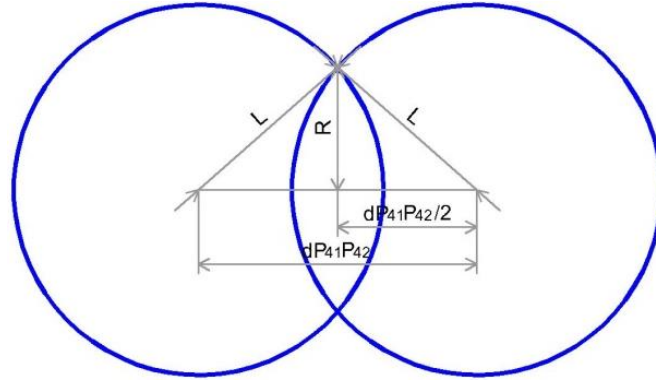
396 **Fig. 14.** Graphic representation where the study of the orientation is developed ( $\beta = 0$ ).

397  $P_{41}$  y  $P_{42}$  will be the centres of the convergence surfaces of the ellipsoid 1 and 2 ( $l = c_2$ ).

398 The red and discontinuous curve will be the convergence curve (a circle).

399

400 The equations that define the position of  $P_{41}$ ,  $P_{71}$ ,  $P_{42}$  y  $P_{72}$  have been already obtained  
 401 in Eq. (10) and Eq. (11). On the other hand, the intersection between two convergence  
 402 surfaces will be the intersection between two spheres with the same radius (Fig. 15).



403  
 404

405 **Fig. 15.** Relationship between the radius of the convergence curve and the  
 406 convergence surfaces.

407

408 Consequently, the radius of the circle will be:

409

$$L^2 = R^2 + \left(\frac{dP_{41}P_{42}}{2}\right)^2 \rightarrow R = \sqrt{L^2 - \left(\frac{dP_{41}P_{42}}{2}\right)^2} \quad (32)$$

410

411 The parametric equation of the convergence curve in local coordinates ( $X_L$ ,  $Y_L$ ,  $Z_L$ ) is:

412

$$x_L(\gamma) = \sqrt{L^2 - \left(\frac{dP_{41}P_{42}}{2}\right)^2} \cdot \cos(\gamma) \quad \text{with } 0^\circ < \gamma < 360^\circ \quad (33)$$

413

$$y_L(\gamma) = \sqrt{L^2 - \left(\frac{dP_{41}P_{42}}{2}\right)^2} \cdot \sin(\gamma) \quad \text{with } 0^\circ < \gamma < 360^\circ \quad (34)$$

414

$$z_L(\gamma) = 0 \quad (35)$$

415

416 The goal is to obtain the equation of the convergence curve in global coordinates ( $X_G$ ,  
 417  $Y_G$ ,  $Z_G$ ). To achieve that,  $Z_L$  must be parallel to the axis of the circle and  $X_L$ ,  $Y_L$  can have  
 418 any orientation due to the infinite planes of symmetry. Also, the orientation of  $X_L$  will  
 419 have a value with the component “Y” equal to 0. This decision will allow the

420 elimination of one of the rotation matrixes in the transformation of a 3D reference  
 421 system. The final expression of the global coordinates is (the development is  
 422 demonstrated in appendices):

423

$$\begin{pmatrix} Pf_{xG} \\ Pf_{yG} \\ Pf_{zG} \end{pmatrix} = \begin{pmatrix} \frac{P_{41X} + P_{42X}}{2} \\ \frac{P_{41Y} + P_{42Y}}{2} \\ \frac{P_{41Z} + P_{42Z}}{2} \end{pmatrix} + \begin{pmatrix} \cos(\alpha_X) & \sin(\alpha_X) \cdot \sin(\alpha_Y) & -\cos(\alpha_Y) \cdot \sin(\alpha_X) \\ 0 & \cos(\alpha_Y) & \sin(\alpha_Y) \\ \sin(\alpha_X) & -\cos(\alpha_X) \cdot \sin(\alpha_Y) & \cos(\alpha_X) \cdot \cos(\alpha_Y) \end{pmatrix} \cdot \begin{pmatrix} x_L(\gamma) \\ y_L(\gamma) \\ z_L(\gamma) \end{pmatrix} \quad (36)$$

424

425 The next steps will be:

426

427 - Step 1: Equations of ellipsoids 1 and 2 in global coordinates.

428 - Step 2: Definition of the lines  $r_1$  and  $r_2$ .

429 - Step 3: Intersection between ellipsoid 1 and  $r_1$  ( $P_a$ ) and between ellipsoid 2 and  $r_2$   
 430 ( $P_b$ ).

431 - Step 4: Line from  $P_{71}$  to  $P_a$  with a length of  $L$  ( $P_c$ ) and line from  $P_{72}$  to  $P_b$  with a length  
 432 of  $L$  ( $P_d$ ).

433 - Step 5: Angle between the vectors  $\overline{P_f P_c}$  and  $\overline{P_f P_d}$  ( $\beta$ ).

434

435 It is important to highlight that the domain of  $t_1$  and  $t_2$  will be:  $(0^\circ, 180^\circ]$  (the value of  
 436  $0^\circ$  is not included because the solution is a line and not a scissor). Also, the following  
 437 study has been developed with  $L = 2 \cdot a_1$  to guarantee the constant value of the focal  
 438 distance.

439

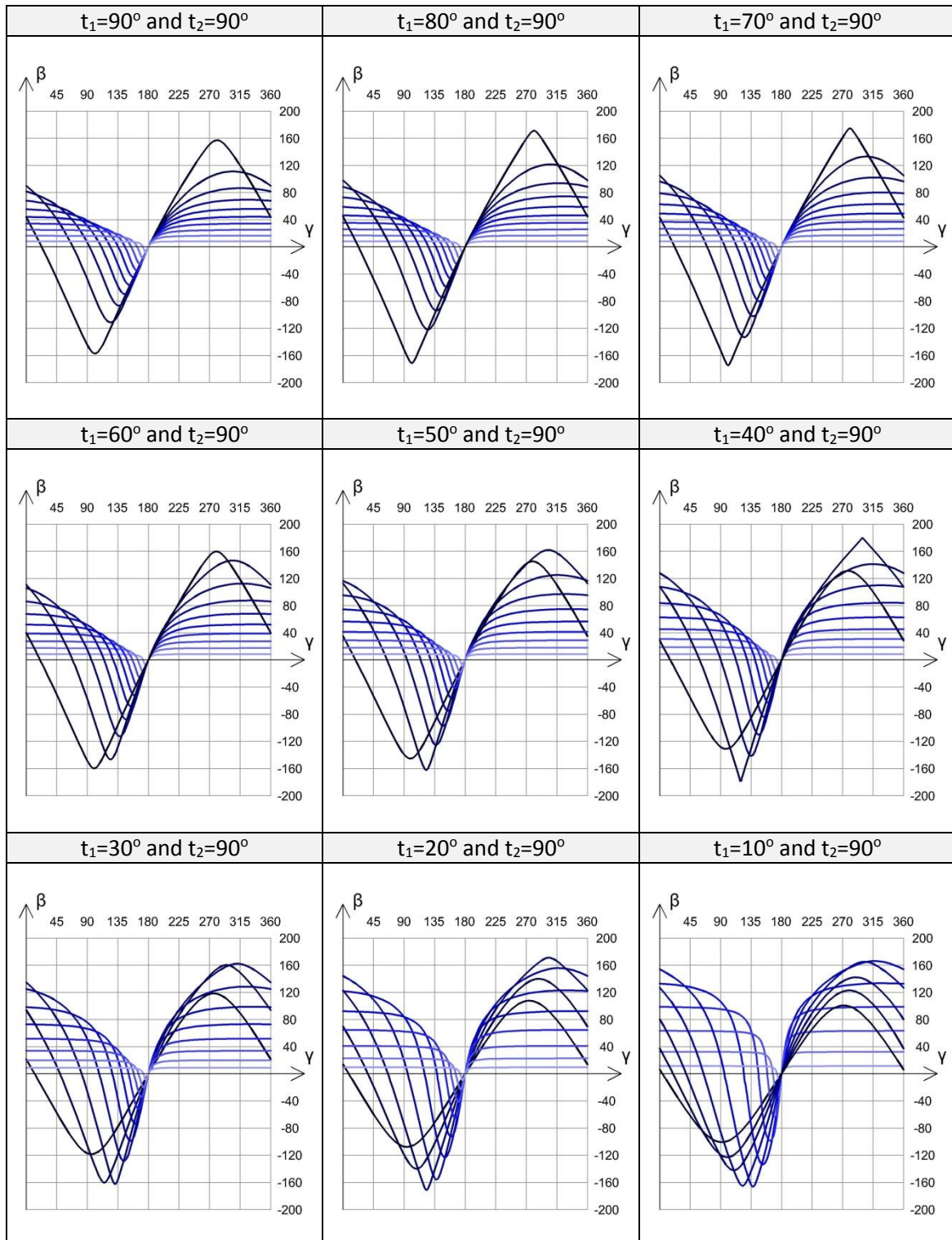


440 4.1. One "t" value is 90° (in this case t<sub>2</sub>) and the other "t" value is iterated from 0° to

441 90°. The values of β and γ for each relationship of b<sub>1</sub>/a<sub>1</sub> have been represented in











442 Table 2:

443



444

445

 $b_1/a_1=1$	 $b_1/a_1=0.9$	 $b_1/a_1=0.8$	 $b_1/a_1=0.7$	 $b_1/a_1=0.6$
 $b_1/a_1=0.5$	 $b_1/a_1=0.4$	 $b_1/a_1=0.3$	 $b_1/a_1=0.2$	 $b_1/a_1=0.1$

446

447

**Table 2.** Evolution of the orientation of the scissors with a value of "t" fixed at 90°.

448

449

The main property of the Table 2 is the number of times that the graphics cut the

450

horizontal axis (the number of times with  $\beta = 0$ ). For these cases, the orientation

451

between ellipsoids 1 and 2 will be the same and there will be a geometric solution. This

452

situation can be represented with the next equation:

453

$$\text{Number of possible solutions} = (\text{Number of times with } \beta = 0) - 1 \quad (1 = \text{trivial solution}) \quad (37)$$

454

455

As can be observed in the graphics of Table 2, all curves cut the horizontal axis twice

456

and there will be only one geometric solution. If the geometric solution is drawn for

457

each graphic, the final scissor module is always a perpendicular extrusion with respect

458

to the plane that contains the scissor between ellipsoids 0 and 1. This situation just

459

allows the design of deployable geometries that are the result of a perpendicular

460

extrusion with respect to their generatrix: planes and cylinders with a simple

461

curvature. It is important to highlight that in case of a flat geometry, the condition of  $L$

462

$= 2 \cdot a_1$  is not necessary because the scissors only have a relationship of a rotation.

463

However, in the case of a cylindrical geometry, the condition of  $L = 2 \cdot a_1$  is mandatory

464

because the relationship between the scissors is more than a rotation. Some examples

465

of flat deployable structures with elements of the same length can be observed in Fig.

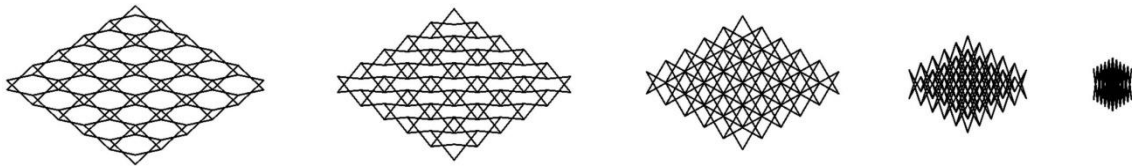
466

16, Fig. 17 and Fig. 18.

467

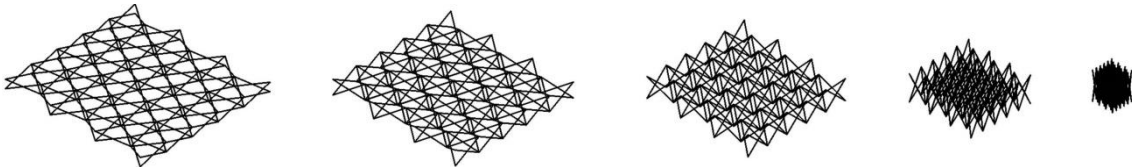
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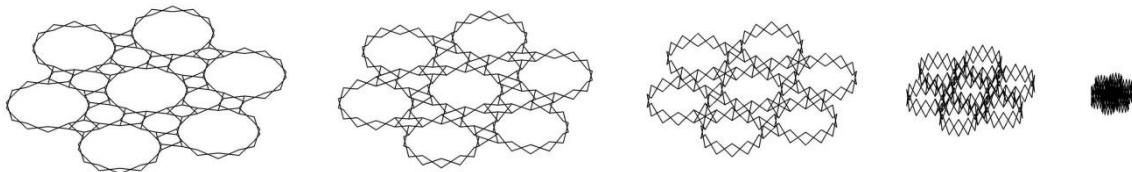
470  
471

472 **Fig. 16.** Flat deployable structure with elements of the same length and with a square  
473 tessellation.



475  
476

477 **Fig. 17.** Flat deployable structure with elements of the same length and with a  
478 triangular tessellation.



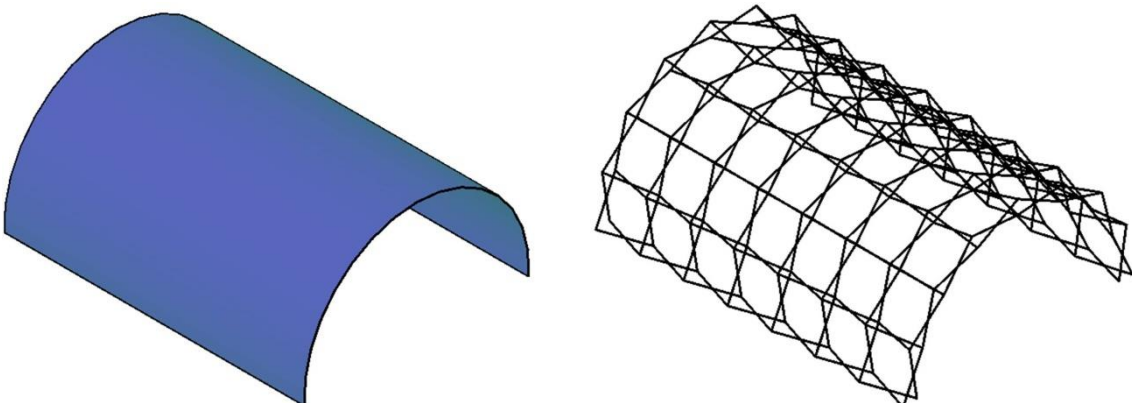
480  
481

482 **Fig. 18.** Flat deployable structure with elements of the same length and with a mixed  
483 tessellation.

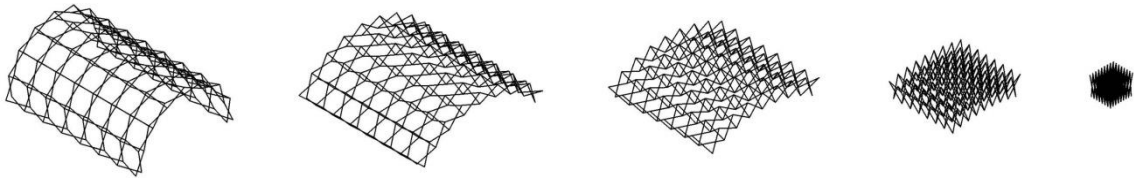
484

485 Some examples of a cylindrical deployable structure with a simple curvature and with  
486 elements of the same length can be observed in Fig. 19 and Fig. 20.

487

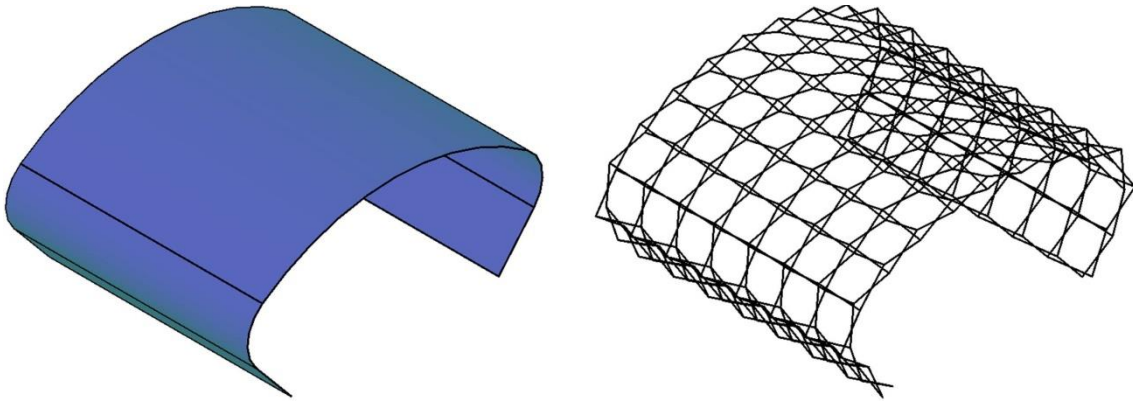


488

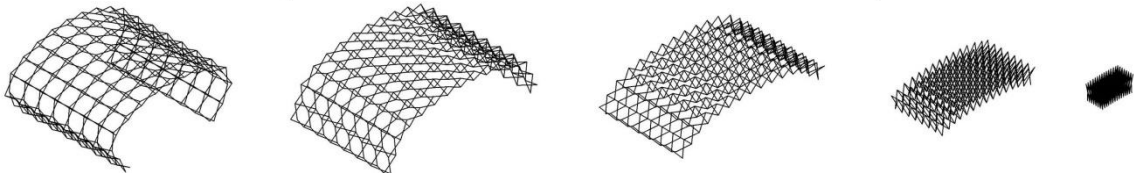


489  
490

491 **Fig. 19.** Cylindrical deployable structure with elements of the same length and with a  
492 circular generatrix.



493



494  
495

496 **Fig. 20.** Cylindrical deployable structure with elements of the same length and with a  
497 mixed generatrix.

498

499

500

501

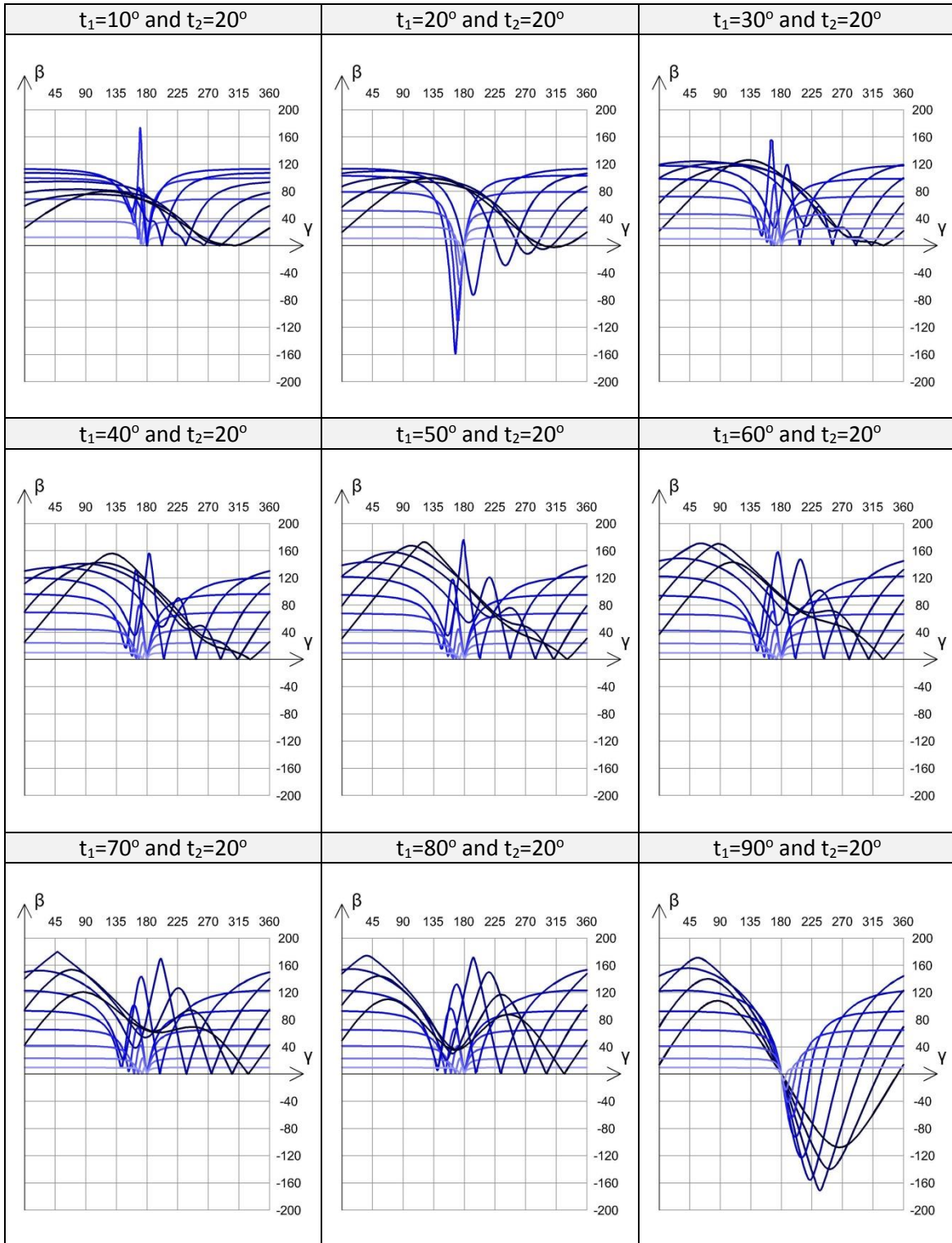
502

503

504

505

506 4.2. One “t” value is different in comparison with 90° (in this case  $t_2 = 20^\circ$ ) and the  
 507 other “t” value is iterated from 0° to 90°. The values of  $\beta$  and  $\gamma$  for each relationship of  
 508  $b_1/a_1$  have been represented in Table 3:



509

$b_1/a_1=1$	$b_1/a_1=0.9$	$b_1/a_1=0.8$	$b_1/a_1=0.7$	$b_1/a_1=0.6$
$b_1/a_1=0.5$	$b_1/a_1=0.4$	$b_1/a_1=0.3$	$b_1/a_1=0.2$	$b_1/a_1=0.1$

510

511 **Table 3.** Evolution of the orientation of the scissors with a value of "t" fixed at 20°.

512

513

514 As can be observed in Table 3, the graphics only cut the horizontal axis two times in

515 two cases: for  $t_1 = t_2$  and for  $t_1 \neq t_2$  but with one value of "t" equal to 90°.

516

517 In the rest of the cases, the graphics only cut the horizontal axis once (the trivial

518 solution). On the other hand, the case of  $t_1 = t_2$  and of  $t_1 \neq t_2$  but with one value of "t"

519 equal to 90° have been already studied in Table 2 (planes and cylinders with a simple

520 curvature).

521

522 Consequently, the only case that shall be analysed is  $t_1 = t_2$ . If both angles ( $t_1$  and  $t_2$ )

523 have the same value, there is symmetry in the original scissors and in the final scissors.

524 This situation means that the curvature of the deployable surface is going to be

525 constant (in the next module, the scissors are going to be the same with the same

526 rotation between them). The only surface that has a constant curvature in all of its

527 points is a sphere. Likewise, this deployable structure with the shape of a sphere must

528 have polar units because it has been demonstrated that only flat deployable structures

529 are possible for  $t_1 = t_2 = 90^\circ$ .

530

531 If the study of Table 3 is done for the interval from 90° to 180°, the number of

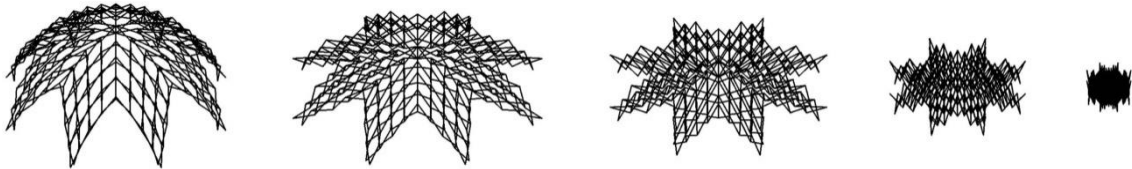
532 intersections with the horizontal axis will be the same but with different positions. This

533 part of the study has been removed to avoid an excessive quantity of tables.

534

535 For  $t_1 = t_2$ , both original scissors are the same and, in consequence, there is a  
536 relationship between them of a rotation (the condition of  $L = 2*a_1$  is not mandatory).  
537 Some application examples can be observed in Fig. 21, Fig. 22 and Fig. 23 using  
538 different relationships between  $L$  and  $a_1$ .

539



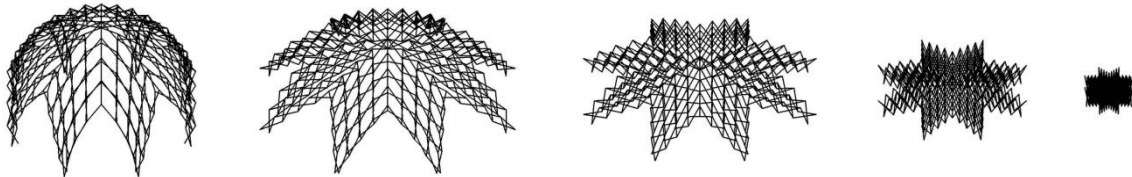
540

541

542 **Fig. 21.** Deployable structure with the shape of a sphere and with all elements of the  
543 same length ( $L = 0.75*2*a_1$ )

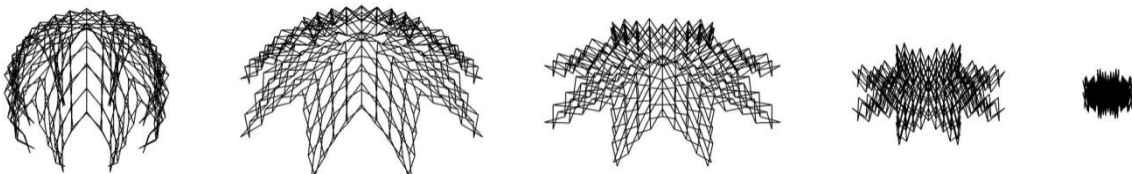
544

545



546 **Fig. 22.** Deployable structure with the shape of a sphere and with all elements of the  
547 same length ( $L = 2*a_1$ )

548



549

550

551 **Fig. 23.** Deployable structure with the shape of a sphere and with all elements of the  
552 same length ( $L = 1.25*2*a_1$ )

552

553

554

555

556

557 **6. Results:**

558

559 The goal of this section is to evaluate the influence of using elements with identical  
560 length in terms of structural behaviour. To achieve that, two spherical deployable  
561 structures are going to be designed where one model (Model A) is going to have all  
562 elements with the same length (length = 1.675 meters) and the other model (Model B)  
563 is going to have elements with a no identical length. The conditions of the simulation  
564 are going to be the next:

565

566 a) Design conditions:

567 - Both models will have a distance between supports of 10 meters.

568 - Both models will have the same number of joints.

569 - Both models will have the same number of elements.

570

571 b) Calculation conditions:

572 - Both models will have the same weight of elements, tendons and textile.

573 Consequently, the price of both models will be the same.

574 - Both models will have the same cross section for the elements, tendons and textile.

575 - Both models will satisfy the maximum vertical displacement expected in the Spanish  
576 regulation of structures against a vertical load of  $1 \text{ kN/m}^2$  applied on the surface of the  
577 structure.

578

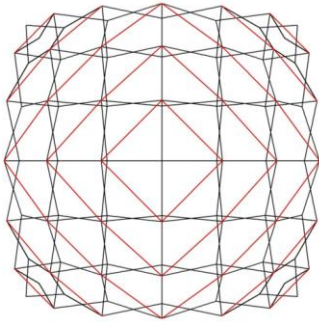
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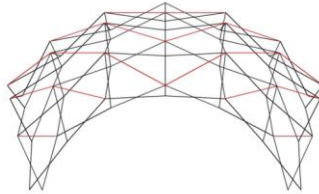
580 Model A (all elements with an identical length) (black colour = elements and red colour  
581 = tendons) (Fig. 24):

582

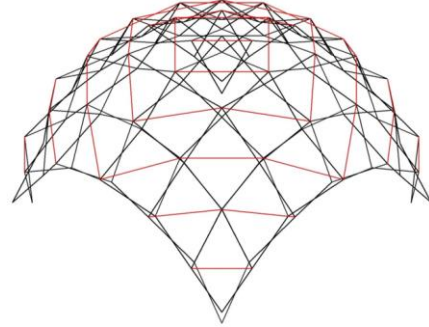
583 a)



b)



c)



584

585

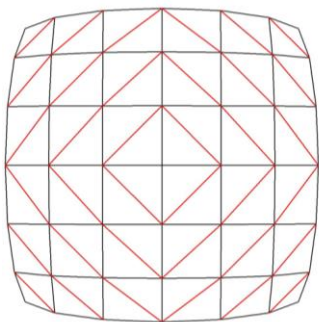
586 **Fig. 24.** (a) Floor view of Model A; (b) Frontal view of Model A; (c) Perspective view of  
587 Model A.

588

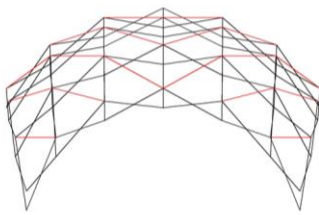
589 Model B (all elements with a different length) (black colour = elements and red colour  
590 = tendons) (Fig. 25):

591

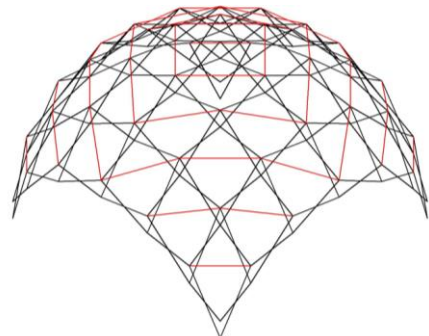
592 a)



b)



c)



593

594

595 **Fig. 25.** (a) Floor view of Model B; (b) Frontal view of Model B; (c) Perspective view of  
596 Model B.

596

597

598

599 6.1. Material and section properties (Table 4 and Table 5):

	Material	Weight per unit volume (kN/m <sup>3</sup> )	Coefficient of Poisson	Modulus of elasticity (kN/m <sup>2</sup> )
Elements	Aluminium	26.6018	0.33	69637055
Tendons	Cable	76.9729	0	1.965 x 10 <sup>8</sup>
Surface	Textile	12.027	0.3	1

600  
601

**Table 4.** Material properties.

	Profile	Outside diameter (cm)	Wall thickness (cm)
Elements	Hollow-circular	6	0.8
Tendons	Solid-circular	1	-
Surface	Shell	-	0.053

602  
603

**Table 5.** Section properties.

604 6.2. Weight and price of each model (Table 6):

	Weight of the elements (kg)	Weight of the tendons (kg)	Weight of the textile (kg)	Price of the structure (joints not included)
Model A	897.66	43.35	55.37 kg	1703.31 \$
Model B	900.13	41.82	54.14 kg	1697.68 \$

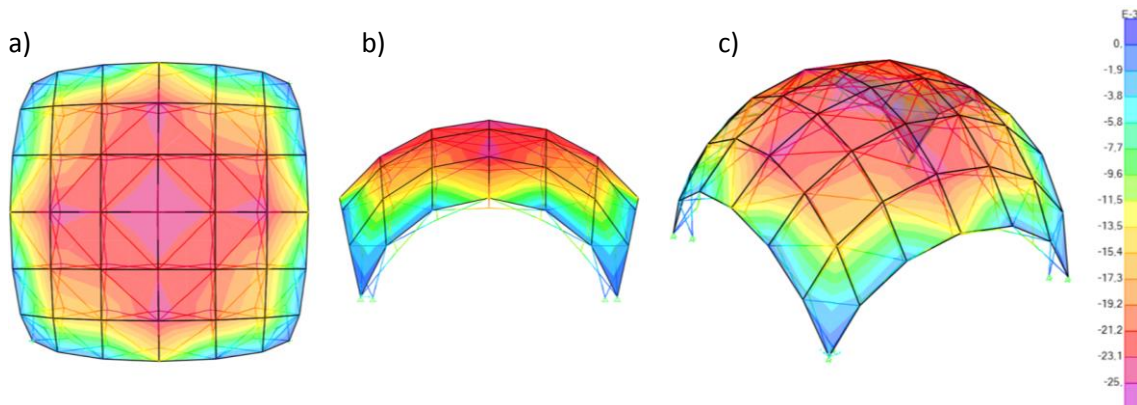
605  
606

**Table 6.** Weight and price of the structure of each model.

607 6.3. Vertical displacements:

608

609 a) Model A (all elements with the same length) (Fig. 26):



610  
611

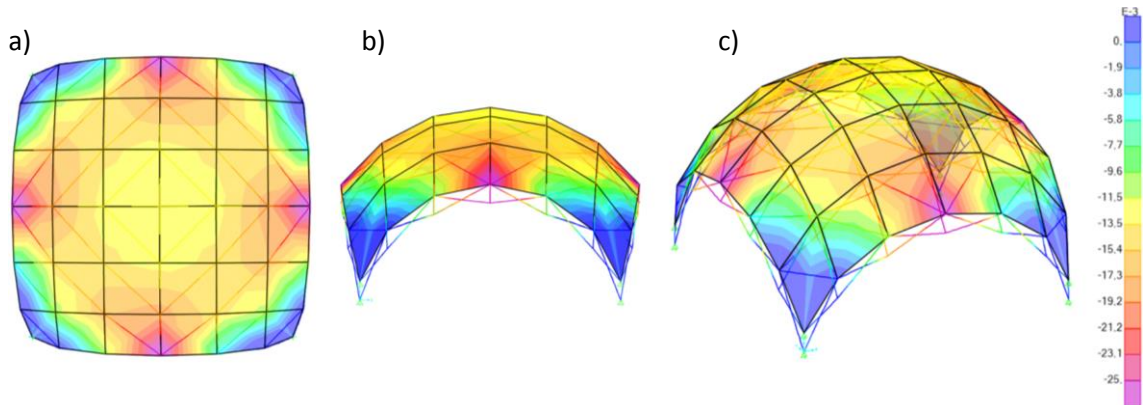
**Fig. 26.** Vertical displacements of the structure with the same length in all elements

612 (scale in meters). (a) Floor view; (b) Frontal view; (c) Perspective view.

613

614 b) Model B (all elements with a different length) (Fig. 27):

615



616

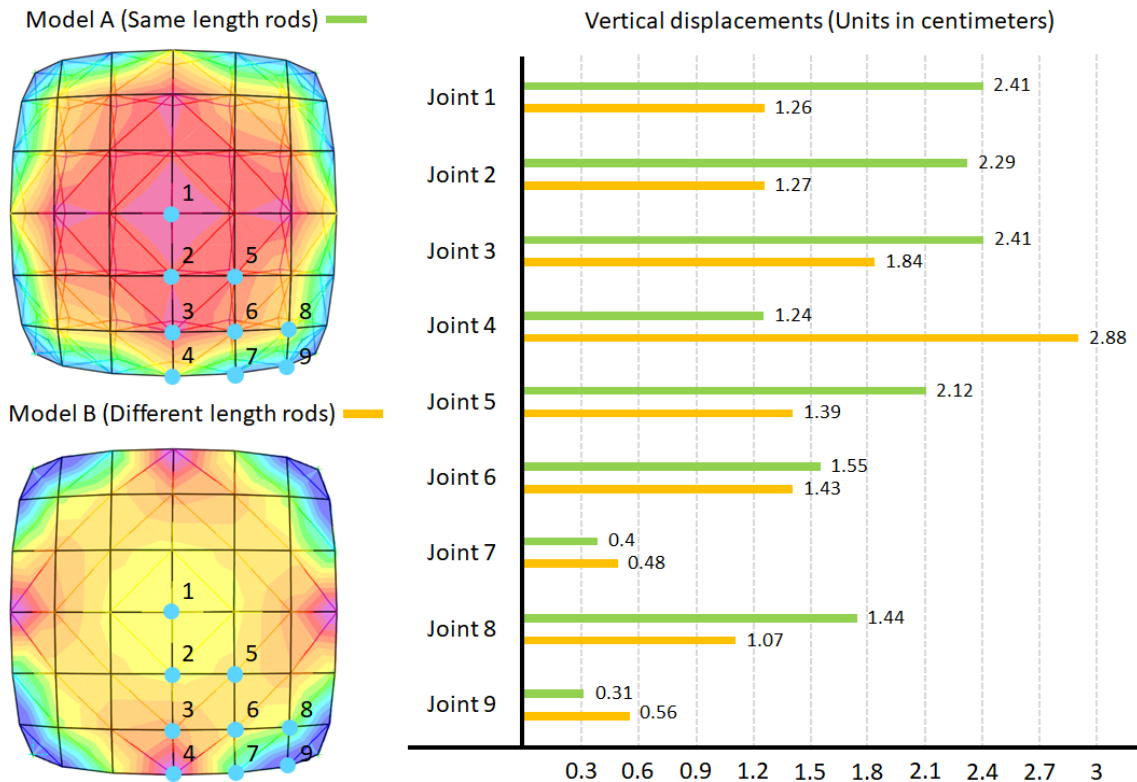
617 **Fig. 27.** Vertical displacements of the structure with a different length in all elements

618 (scale in meters). (a) Floor view; (b) Frontal view; (c) Perspective view.

619

620 The comparison of the vertical displacements is represented in Figure 28:

621



622

623

624 **Fig. 28.** Comparison of vertical displacements between both models (superior joints).

625

626 In addition, the limit of the vertical displacements in function of the Spanish regulation  
627 for structures is:

628 
$$\frac{2 \cdot \text{Distance between point A and point B}}{3} \geq \text{Vert. displac. of point A} - \text{Vert. displac. of point B} \quad (38)$$

629

630 The previous equation must be satisfied for all possible combinations of the points of  
631 the structure. The worst combination for the Model A is: Point A = Joint 6; Point B =  
632 Joint 7 and distance between point A and point B = 1.75 meters

633 Consequently:

634 
$$\frac{2 \cdot 1.75}{3} \geq 1.55 - 0.4 \rightarrow 1.16 \geq 1.15 \quad (39)$$

635

636 The worst combination for the Model B is: Point A = Joint 4; Point B = Joint 3 and  
637 distance between point A and point B = 1.59 meters

638 Consequently:

639 
$$\frac{2 \cdot 1.59}{3} \geq 2.88 - 1.84 \rightarrow 1.06 \geq 1.04 \quad (40)$$

640 As can be observed in Figure 28, the behaviour of the Model A (all elements with the  
641 same length) in terms of vertical displacements is worse in comparison with the Model  
642 B (all elements with a different length): only 3 points (joints 4, 7 and 9) have a lower  
643 vertical deformation in the Model A. In conclusion, the vertical displacements of a  
644 spherical deployable structure using the same length in all elements are worse in  
645 comparison with the use of elements with a different length.

646

647

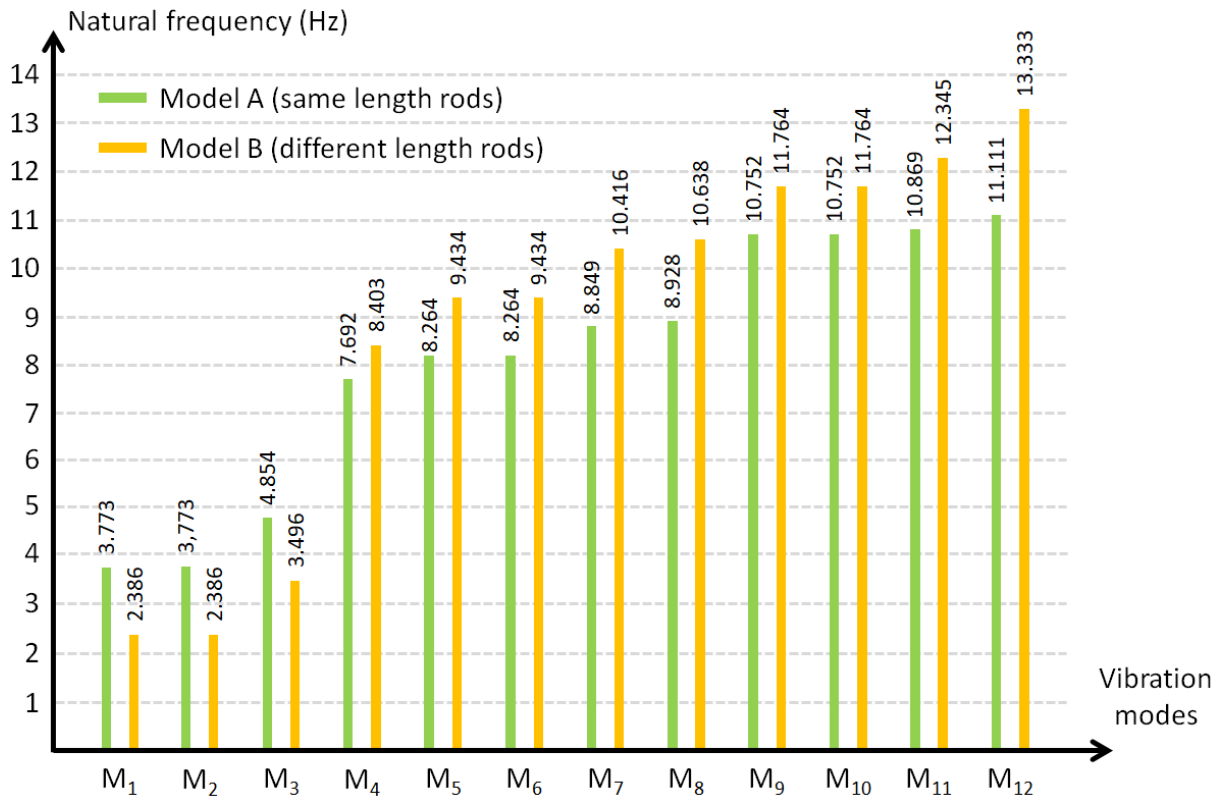
648

649

650 6.4. Natural frequencies:

651

652 The natural frequencies of both models are:



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654

655

**Fig. 29.** Natural frequencies and vibration modes for each model.

656

When the rigidity of a structure is evaluated, M<sub>1</sub>, M<sub>2</sub> and M<sub>3</sub> are always the most important vibration modes and, as can be observed in Figure 29, the Model A has a better value of the natural frequencies in M<sub>1</sub>, M<sub>2</sub> and M<sub>3</sub> (the higher is the natural frequency, the higher is the rigidity of the structure).

660

661

Therefore, the behaviour of Model A against the loss of stiffness is approximately 40% - 60% better than Model B and, in consequence, the rigidity of a deployable structure using the same length in all elements is better in comparison with the use of elements with a different length.

665

666 **7. Conclusions:**

667

668 The design possibilities and the structural behaviour of deployable structures with  
669 identical elements is a topic that had been never researched in deep. The results of  
670 this paper give an overview of the geometries that can be developed using identical  
671 elements: flat shapes, cylinders with any generatrix, spheres and combinations of  
672 these options. Furthermore, the use of this geometric constraint allows the creation of  
673 deployable structures with higher vertical deformations and natural frequencies  
674 (better stability of the structure against horizontal displacements). Having these  
675 parameters in consideration, the decision of using identical elements will be based on  
676 the requirements of the structure in terms of the geometric complexity and the  
677 structural regulation of the country.

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691

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